

# Hydrogen Combustion Research for Gas Turbine Engines

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**IEA Task Leaders Meeting  
August 13, 2006**

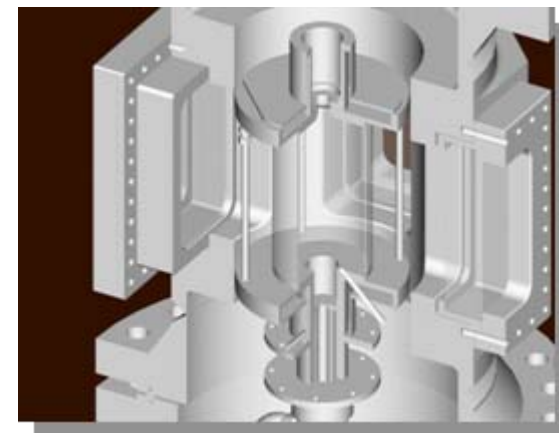


# Partnering with National Energy Technology Laboratory (NETL)



## Approach:

- Extend data base for lean premixed swirl burners to realistic pressures and temperature. Emphasis on H<sub>2</sub>-enriched fuels.
- Atmospheric-pressure tests in SimVal burner at Sandia; high-pressure tests at NETL.
- Utilize Sandia's diagnostic expertise for the development of high pressure diagnostics in realistic gas turbine environments.



## Sandia Burner

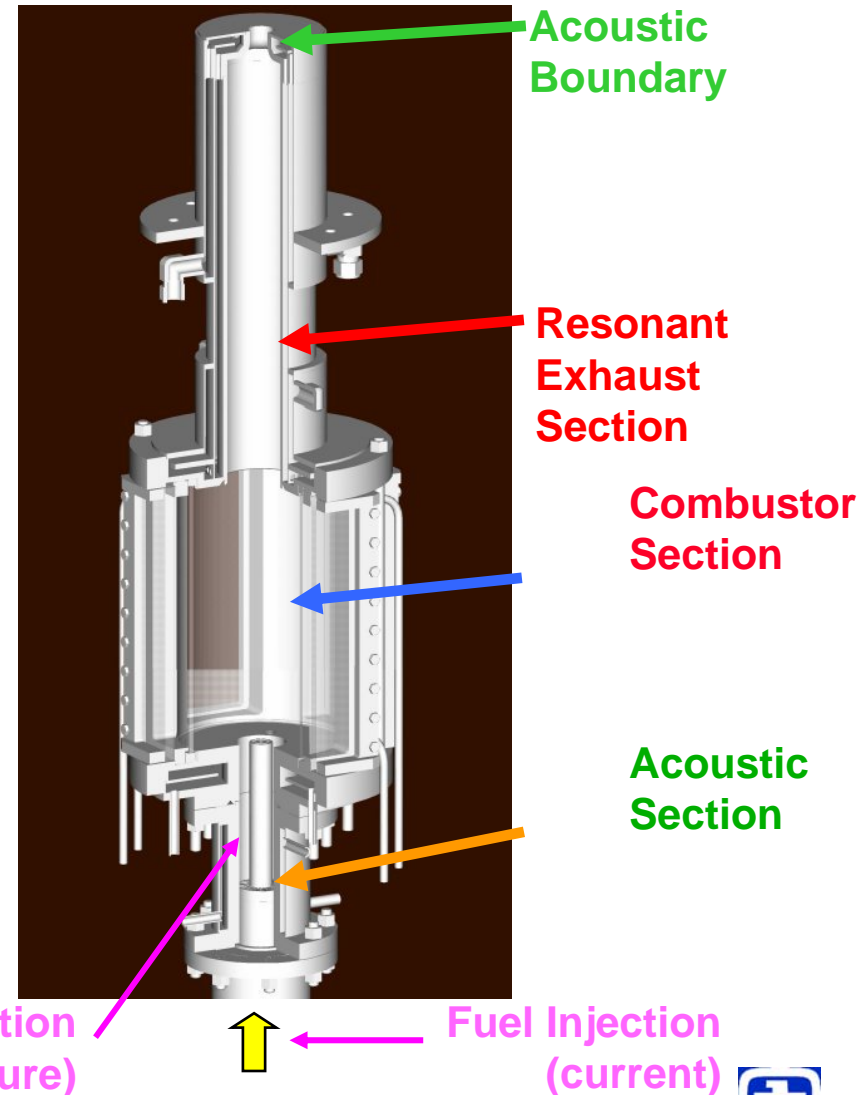
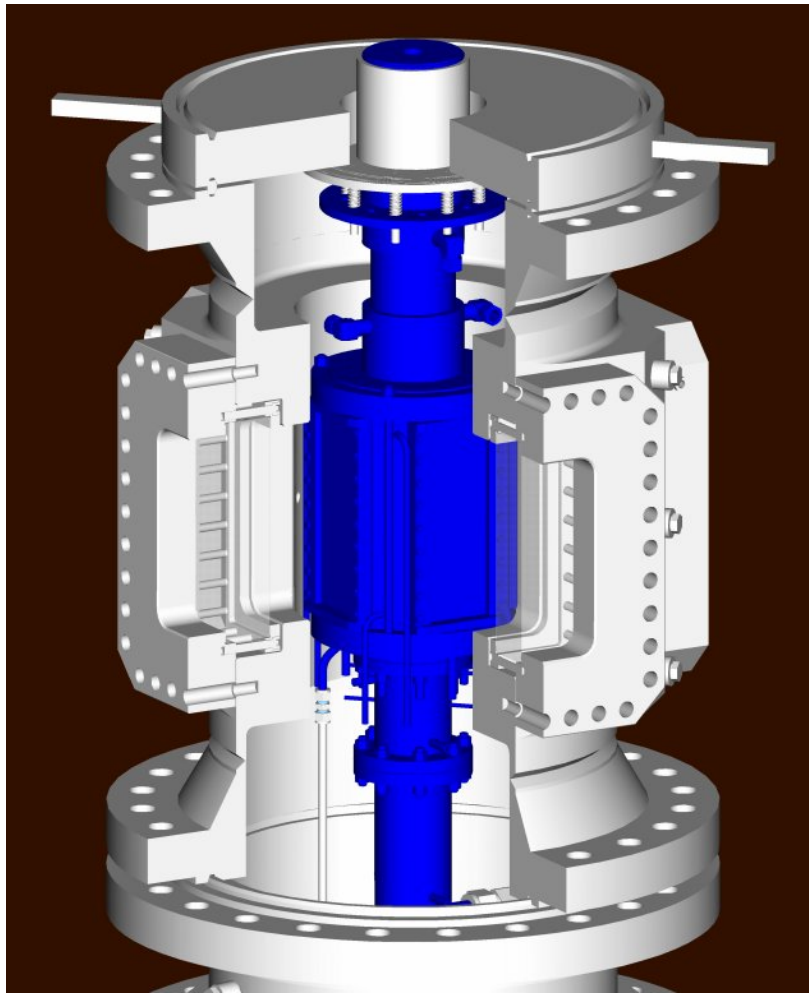
- Atmospheric pressure operation
- Design optimized for Sandia CRF laboratory facilities
- Full optical access for optimal use advanced diagnostics



## NETL Burner

- Operation to 30 atmospheres
- Inlet temperature to 800 K
- Optical access
- Limited datasets at elevated pressure

# SimVal Combustor Geometry



# *SimVal Project Goals*



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- ⇒ **Provide data sets for the validation of CFD (LES) simulations to aid the development of advanced gas turbine combustors**
    - **Gather data at elevated combustor pressures**
    - **Emissions**
    - **Dynamic modes and pressures**
    - **Flow field characterization**
    - **Boundary condition characterization**
    - **Dynamic events and transitions**
      - **Lean Blow-Off**
      - **Flashback**
      - **Abrupt changes in dynamics and emissions**
    - **Effects of fuel variability on emissions, dynamics, lean blow-off, and flashback.**

# *SimVal Facility Test Capabilities*



## ⇒ Combustion Air

- Flow Rate: 0 - 70,000 slm (0 – 1.1 kg/s)
- Preheat Temperature: 0 – 800 °K

## ⇒ Natural Gas

- Flow Rate: 0 - 4,700 slm

## ⇒ Hydrogen

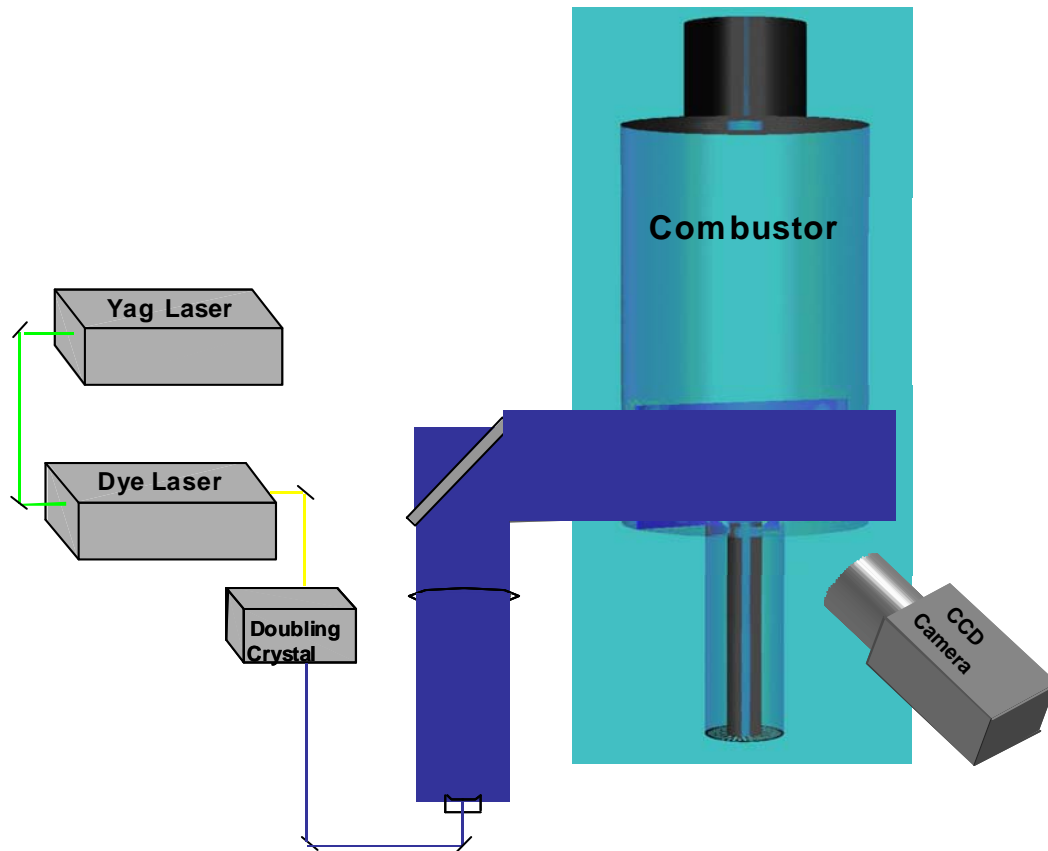
- Hydrogen flow rate: 0 - 470 slm

## ⇒ Test Section Operating Conditions

- Pressure: 0 - 2,200 kPa (22 atm)
- Temperature: 0 – 800 °K



# Experimental System



## Test Conditions

Pressure 1-8 atm

Preheat 520-600 K

H<sub>2</sub> addition 0-60% in NG

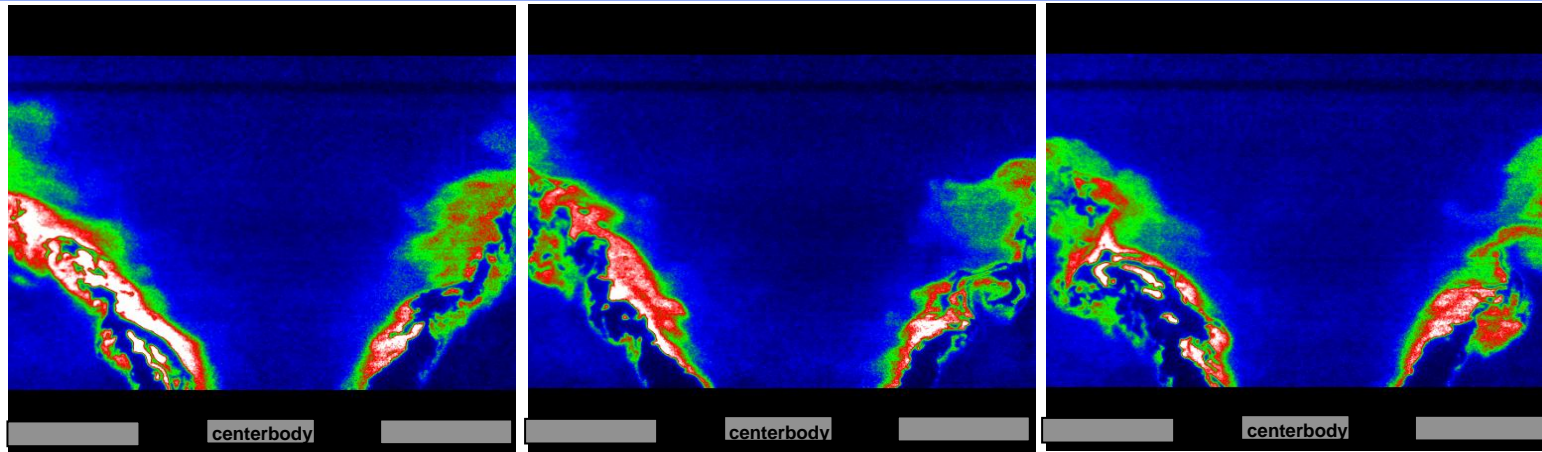
Equivalence ratio 0.5-0.6



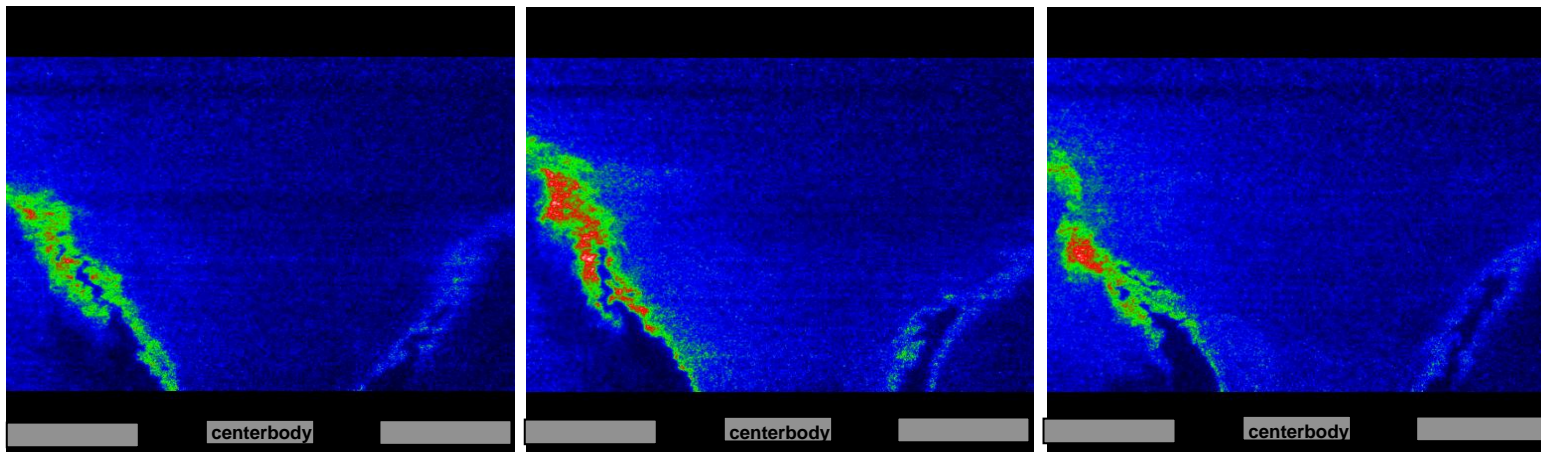
# Effect of Pressure



60% H<sub>2</sub> / 40% NG  $\Phi=0.6$   $T_{in}=522-580K$   $V=40$  m/s



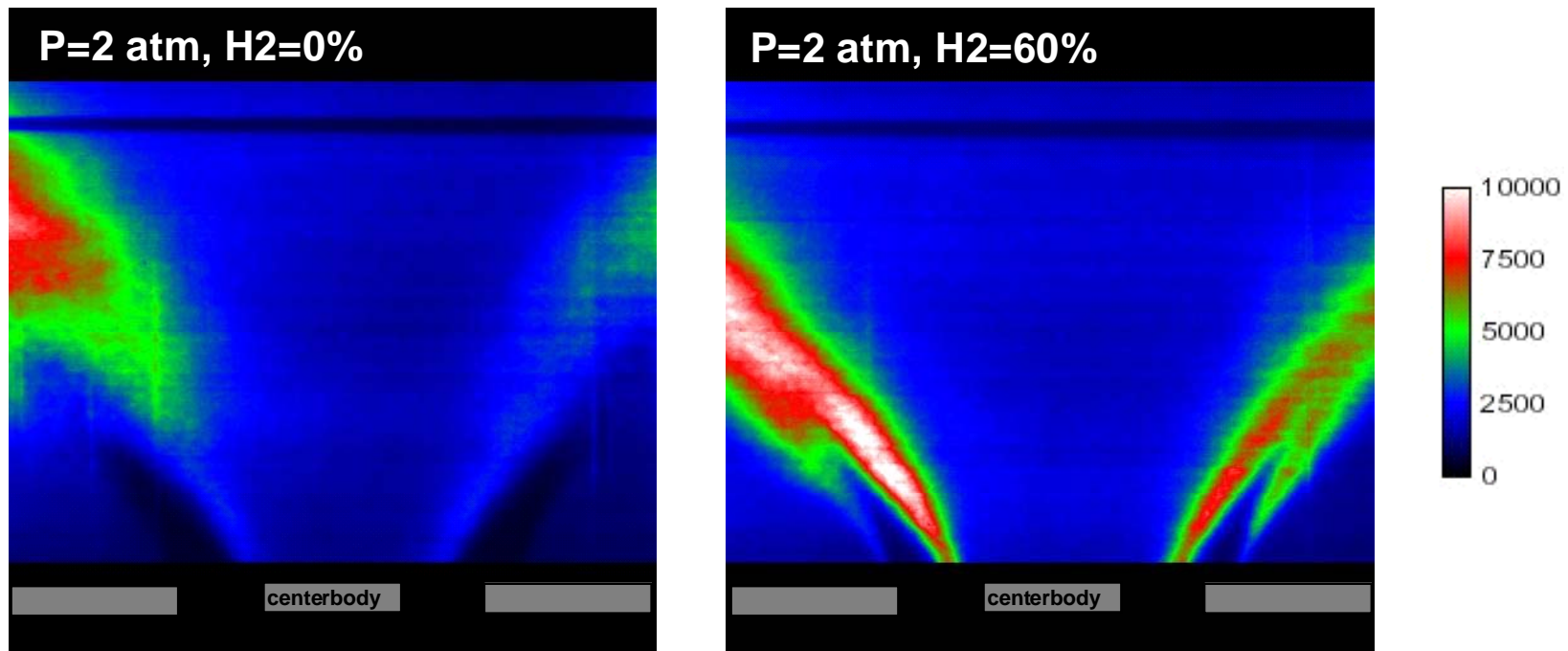
**P=1atm**  
**Scaled**  
**0-20,000**



**P=8atm**  
**(50% H<sub>2</sub>)**  
**Scaled**  
**0-5,000**

- Increased pressure reduces flame thickness

# *Effect of H<sub>2</sub> Addition*



- H<sub>2</sub> addition reduces flame thickness and moves flame stabilization point farther upstream